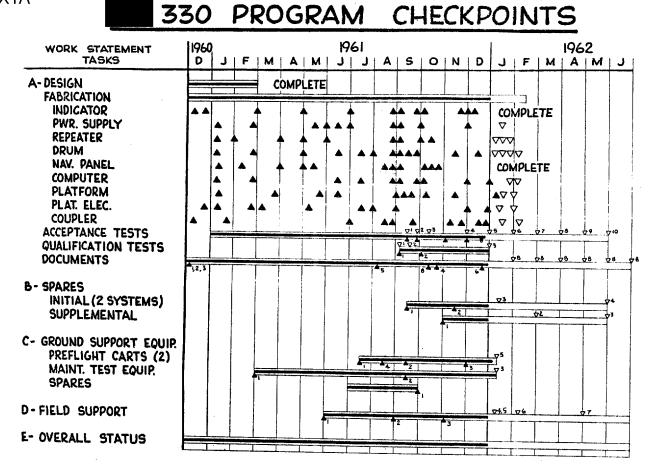
MH 330



**JANUARY 4, 1962** 

25X1A

4 JANUARY 1962



25X1A

330 Checkpoints

	Task	Date				
Α.	Design Fabrication Acceptance Tests	Complete As Shown Systems Complete as Shown				
	Qual. Tests					
	<ol> <li>Start Subsystem Tests</li> <li>Start System Tests</li> <li>Qual. Tests Complete</li> </ol>	8-21-61 9-15-61 1-1-62				
,	Documents					
B•	1. Preliminary Installation Layout 2. Preliminary Interconnecting Diagram 3. Final Interconnecting Diagram 4. Final Performance Spec. 5. Final Installation Layout 6. Technical Manual 7. Technical Progress Letter 8. Acceptance Test Report  Spares	6-1-60 6-1-60 12-1-60 10-20-61 8-4-61 12-20-61 Monthly 1 Month Following Delivery of each system				
	Initial	2				
	1. First Set Completed 2. Second Set Completed 3. First Set Delivered 4. Second Set Delivered	9-15-61 11-15-61 1-11-62 . 4-24-62				
	Supplemental					
	<ol> <li>Go Ahead</li> <li>Fabrication Complete</li> <li>Delivery</li> </ol>	11-1-61 3-1-62 6-1-62				

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330 Program Checkpoints
Page Two

### C. GSE

### Preflight Carts

2. 3. 4.	First Cart Completed First Cart Delivered Mods Complete, First Cart Second Cart Complete Second Cart Delivery	7-15-61 9-15-61 Complete 8-15-61 1-11-62
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## Maintenance Test Equipment

1.	Fabrication Complete	7 7 / 2
2.	Testing Complete	3-1-61
	Delivery	9-15-61
-	2011/01/	1_11_62

### Spares

1.	Fabricated,	In	Stock		10-15-61
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### D. Field Support

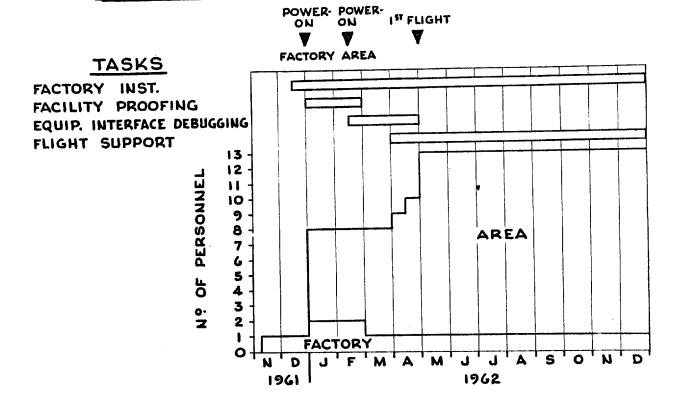
5. 5.	First Field Support Rep. In Plant Assigned Field Support Manager First Rep. to Factory Transfer of Personnel to Area Power-On Area First Flight INS Flight	6-1-61 9-1-61 11-5-61 1-7-62 1-7-62 2-7-62
7.	INS Flight	5-1-62

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AS OF 1 OCT. 1961



# 330 TECH. REP. SUPPORT



4 JANUARY 1962

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# 330 DELIVERY STATUS

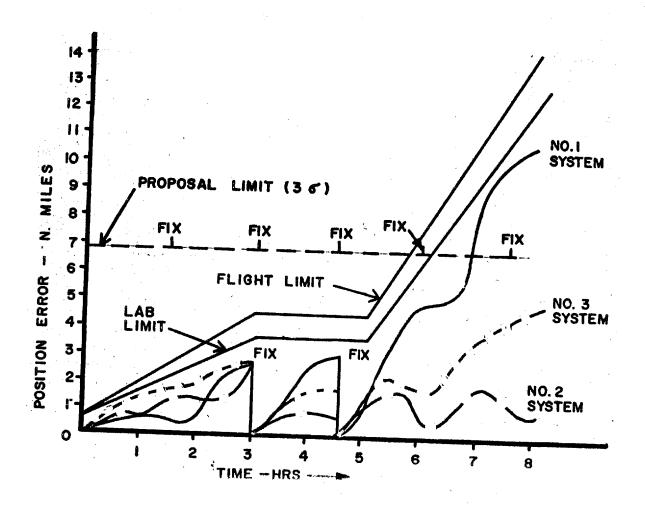
ITEMS OF	1961			1962												
DELEVERY	5	0	N	D	J	F	М	Д	М	J	J	A	5	0	N	D
SYSTEM J-1	E	FAC	TORY	2	<b>У</b> '	<b>∇3</b>			74							
J-3 J-4 J-5 J-6 J-7 J-8 J-9					∇	. ∇,	∇ <sub>i</sub>	<b>⊘</b> i ▽	$\triangle'$	<b>↓</b>						
PREFLIGHT NO.1 CARTS NO.2	Ę	FAC.	TORY J		- -   ∇ <sub>i</sub>					$\nabla$						
MAINT: TEST EQUIP.  COMPUTER TEST UNIT  DRUM LOADER  PLAT. ELECT. TEST STA.  COUPLER TEST STA.  RECORDER RACK					▷         ▷         ▷         ○ <t< td=""><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td></t<>											

Approved For Release 2001/03/01 : CIA-RDP33-02415A000500390017-5

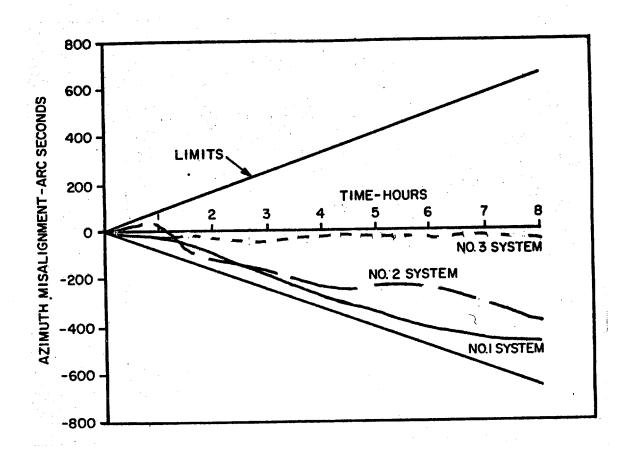
# 25X1A Approved For Release 1493/01; CIA-RDP33-02415A000500390017-5

Items of Delivery	System Number	Checkpoints	Date
System	J-1 J-2	1 1 2 3 4	1-11-62 9-15-61 (Delivered) to Factory 1-7-62 To Area 2-7-62 First Flight 5-1-62 First INS
	J-3	1	Flight 1-18-62 For Flight Test
	J-4	1	2-21-62
•	J-5	ı	3-15-62
	J-6	1	4-7-62
	J-7	1	4-28-62
	J-8	1	5-15-62
	J-9	1	6-7-62
	J-10	1	6-28-62
Preflight Carts #1		1	9-15-61 (Delivered)
#2		1	1-11-62
Maintenance Test Equipment	ment		
Computer Test Unit		1	1-11-62
Drum Loader		ĺ	1-11-62
Platform Elect. Tes		ī	1-11-62
Coupler Test Static	on	ī	1-11-62
Recorder Rack	•	1	1-11-62

MAY APRIL MARCH 1962 PROPOSED FLIGHT TEST SCHEDULE INS FEBRUARY JANUARY POLAR MILESTONE FLIGHTS ZI MILESTONE FLIGHTS INSTALLATION SUPPORT TASK PERFORMANCE FLIGHTS GROUND TEST



Position Error Test Results



Azimuth Error Test Results

	FAILURES	0		0		0	SEE VIBRATION CHART	
330 QUALIFICATION TEST STATUS	COMPLETE	100%	aates 100%	100%	100%	100%	\$08°	
	TEST	POWER VARIATION	FREQUENCY RESPONSE AND ANGULAR RATES	CORONA-ALTITUDE	SHOCK	HUMIDITY - TEMPERATURE	VIBRATION	

\* ONLY RETESTING OF FIXES REMAINING

	3	
ACTIVATE OF CAME	10 T T T T T T T T T T T T T T T T T T T	
7 CEE	277	

Ар	pro	ved	For	Releas	se 2	001/ <u>\$</u> 3/0	1 မ္မ်ိဳCIA-	RDP33-0
FIX	Additional Tie-downs	Edditional Tie-downs		Additional Tie-downs & Redesign of Transformer		Stiffen boards, add addi- tional support, vibration all production boards.	Ada base plate and stiffen on covers.	Change shock mount. Imposed tighter qual. control on cgimbal bearings.
FAILURE	Wire breakage	Wire Breakage	None	Wire Breakage & Transformer Shorting	None	Excessive Board Resonance causing open circuits.	Excessive deflection of entire structure causing open circuits.	Gimbal resonance causing gyro drift in some platforms
FIX RETESTING	100%	100%	F (* 42 B)	70%	1 1	85%	95%	50%
TESTING	100%	100%	100%	100%	100%	100%	100%	100%
COMPONENT	POWER SUPPLY	NAV Panel	DTG/GS	REPEATER	PLAT. ELEC.	Computer	COUPLER	PLATFORM

# PROPOSED FLIGHT DATA PROCESSER & RECORDER

Recorded Quantities -

Camera Pitch, Camera Roll, Camera Heading, Drift Angle, Ground Speed, Relative Time. Latitude, Longitude, Ground Angular Rate,

Form of Inputs

and square wave precision frequency. synchros, resolvers, Stepping motors,

Form of Output

All others - Serial Binary whole number sampled once per camera exposure. Time - Precision 400 cps continuous recording

Installation

Weight - 29 lbs.

Power - 400 cps and 28 V DC

Recording time - 4 hours

Cooling Air = 1/2 to 1 lb. depending on temperature. Location - air conditioning bay above INS